

Adhesion Between Thin Cylindrical Shells With Parallel Axes

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Energy principles are used to investigate the adhesion of two parallel thin cylindrical shells under external compressive and tensile loads. The total energy of the system is found by adding the strain energy of the deformed cylinder, the potential energy of the external load, and the surface energy of the adhesion interface. The elastic solution is obtained by linear elastic plate theory and a thermodynamic energy balance, and is capable of portraying the measurable quantities of external load, stack height, contact arc length, and deformed profile in the reversible process of loading-adhesion and unloading-delamination. Several worked examples are given as illustrations. A limiting case of adhering identical cylinders is shown to be consistent with recent model constructed by Tang et al. Such results are of particular importance in modeling the aggregation of heterogeneous carbon nanotubes or cylindrical cells, where the contacting microstructures have a different radius and/or bending stiffness.

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1 Introduction

The adhesion of thin-walled micro- and nanoscale structures governs the functionality of many emerging technologies [1,2]. Fabrication methods in nanotechnology include adhesion-controlled manipulation and assembly of thin-walled structures such as carbon nanotubes (CNTs) single fibers and bundles, graphene sheets, and fullerenes [3]. Thin-film/thin-wall adhesion also controls the stability and structural integrity of flexible nano-electronics and microtruss structures, which are subject to stiction and potential collapse under environment induced adhesion (e.g., meniscus formation at high relative humidity) [4,5]. Moreover, there has been growing interest in the role of thin-walled adhesion in biological and pathophysiological systems. Waste water treatment relies on the adhesion-controlled aggregation of bacteria, and the formation of biofilm [6] and cell-cell adhesion helps form natural and prosthetic tissues [7]. Excessive adhesion causes monocytes to bond to the aorta wall, which eventually obstructs the vessels and leads to atherosclerotic plaques [8], whereas lack of adhesion results in the loss of synaptic contacts and gives rise to Alzheimer disease [9].

Developing insights and predictive models for these systems requires an understanding of the mechanics of adhesion between thin-walled structures as a result of intersurface forces such as electrostatic, van der Waals interactions, and meniscus. To achieve mechanical equilibrium, the adhesion energy must balance the mechanical energies due to external load and structural deformation [10]. Notwithstanding the many existing and successful solid-solid adhesion models, a new theory is needed to explicitly address adhesion between thin-walled structures that are *dissimilar* in stiffness, geometry, and dimension. Here, we consider one particular class of geometries: parallel, thin-walled cylinders with dissimilar bending rigidity and radius. The new model has the potential to be extended to other geometries, such as contacting circular plates and thin-walled spheres [11].

Virtually all existing adhesion models are based on the Hertz contact theory. Because of geometrical incompatibility, exerting an external load on two noninteracting spheres leads to a compressive stress within the contact circle. Modifications to include interfacial adhesion were later introduced by Johnson-Kendall-Roberts (JKR), Derjaguin-Muller-Toporov (DMT), and Dugdale-Barenblatt-Maugis [10]. In essence, the interfacial attraction modifies the local deformation and introduces a tensile stress around the largely compressive contact circle. Relationships between applied load, contact radius, and approach distance are verified in a wide range of materials and interfaces. The theory is further extended to the adhesion of a solid sphere with a wavy substrate [12,13], a solid cylinder with a planar substrate, and cylinders with parallel axes [14,15]. However, these models are inadequate for thin shells in that the shell conforms to the substrate geometry by deforming in plate-bending, membrane-stretching, or mixed bending-stretching mode such that the notion of central compression is excluded. New models are recently developed for freestanding planar circular membranes clamped at the periphery and a planar substrate in the presence of finite range intersurface attraction, though membrane deformation is constrained to membrane stretching and negligible bending [4,16–20].

Thin shell adhesion on a planar substrate has been investigated extensively with numerical methods. Seifert [21] treated lipid vesicles as shells, developed a mechanical model by balancing the adhesion energy with Helfrich's elastic bending, and constructed a self-consistent theory for bounded and unbounded vesicles. Tang et al. [22] and Glassmaker and Hui [23] constructed an elastic model for two interacting CNTs that was consistent with molecular mechanics simulation. A critical shell radius is found below which the contact remains a line: $R_{\min} = (k/\gamma)^{1/2}$, where k is the shell stiffness and γ is the adhesion energy. Adams, Pamp, and Majidi introduced the moment-discontinuity-method to analyze the adhesion of intrinsically curved plates and beams to curved substrates [24,25]. Springman and Bassani [26,27] adopted a numerical method to probe a spherical capped shell attracted to a planar substrate via a finite range Lennard-Jones potential, derived the "pull-in" and "pull-off" events, and further extended their model to wavy substrates under coupled chemomechanical interactions.

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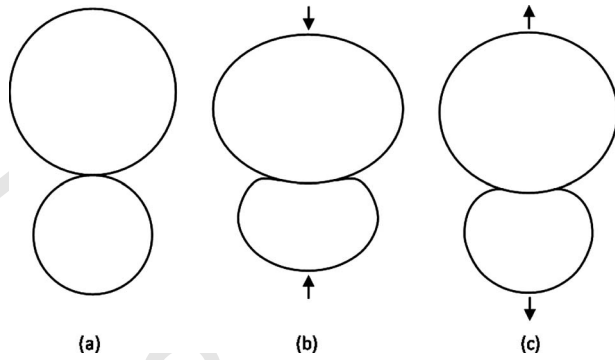


Fig. 1 Interactions between two cylindrical shells: (a) touching at a line contact without adhesion, (b) compressive deformation with adhesion, and (c) tensile deformation with adhesion

76 In this paper, we attempt to address the global deformation of
 77 two elastic cylinders with parallel axis under the following as-
 78 sumptions: (i) Both cylinders are hollow shells with infinite
 79 length, (ii) bending is the dominant deformation mode, and (iii)
 80 the intersurface attraction is effective at intimate contact conform-
 81 ing to the JKR assumption [28]. A boundary condition is intro-
 82 duced to represent the discontinuity in bending curvature at the
 83 contact edge. This is an extension of the moment-discontinuity-
 84 method [26,27] and is derived by minimizing the total potential
 85 energy of the system with respect to the width or radius of the
 86 contact zone. This boundary condition may also be derived using
 87 methods of fracture mechanics such as the J -integral [23,29] and
 88 the stress intensity factor [30]. However, in contrast to the current
 89 analysis, these derivations are beyond the scope of conventional
 90 plate and shell theory and require the evaluation of internal stress
 91 and strain fields.

92 2 Model

93 Figure 1 shows two cylinders with natural undeformed radii R_1
 94 and R_2 being pressed into contact and then separated. Figure 2
 95 shows the curvilinear coordinates. Upon a compressive force F ,
 96 the cylinders deform to create a finite contact segment of arc
 97 length $2a$. As F becomes tensile (negative), adhesion contact re-
 98 mains until a critical pull-off load F_0 is reached. A spontaneous
 99 separation of the adherends follows that reduces a to zero.

100 Let s_1 and s_2 denote the arc lengths of the bottom and top
 101 cylinders, respectively, measured from the cylinder poles. Sym-
 102 metry about the vertical axis requires the left-half of the system to
 103 be considered, and analysis is limited to $L_1 = \pi R_1$ and $L_2 = \pi R_2$.
 104 Define

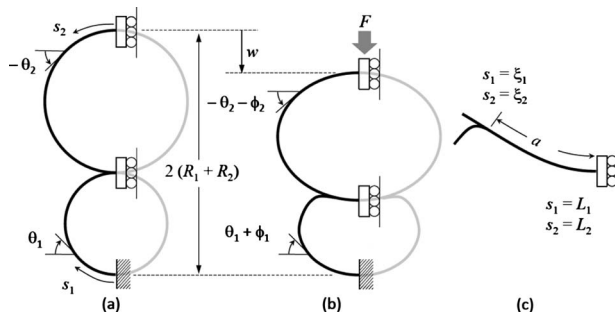


Fig. 2 Curvilinear coordinates

$$\xi_1 = L_1 - a \quad \text{and} \quad \xi_2 = L_2 - a \quad (1) \quad 105$$

106 corresponding to the arc length at which the bottom and top cyl-
 107 inders make contact. In their natural configuration, the cylinders
 108 are deflected by an angle

$$\theta_1 = s_1/R_1 \quad \text{and} \quad \theta_2 = -s_2/R_2 \quad (2) \quad 109$$

110 with respect to horizontal. Under an applied load, the deflection
 111 increases by an angle ϕ_1 and ϕ_2 such that the final deflection is
 112 $\theta_1 + \phi_1$ and $\theta_2 + \phi_2$.

2.1 Boundary Conditions. The angular deformations ϕ_1 113
 114 $= \phi_1(s_1)$ and $\phi_2 = \phi_2(s_2)$ and arc length a must satisfy boundary
 115 conditions that ensure both mirror symmetry about the vertical
 116 axis and geometric compatibility between the cylinders along their
 117 contact. Noting that $\theta_1(0) = \theta_2(0) = \theta_1(L_1) = \theta_2(L_2) = 0$, it follows
 118 that in order for symmetry to be preserved, the boundary condi-
 119 tions

$$\phi_1(0) = \phi_2(0) = \phi_1(L_1) = \phi_2(L_2) = 0 \quad (3) \quad 120$$

121 must be satisfied. To ensure geometric compatibility and to pre-
 122 vent interpenetration of the adhering surfaces, the two cylinders
 123 must share the same shape along the length of contact. Referring
 124 to Fig. 2(b), this requires $\pi - (\theta_1 + \phi_1)$ to equal $-(\theta_2 + \phi_2) - \pi$ for
 125 all values of $s_1 \in [\xi_1, L_1]$ and $s_2 \in [\xi_2, L_2]$, where $s_2 = s_1 - \xi_1 + \xi_2$,

$$\theta_1(s_1) + \phi_1(s_1) = 2\pi + \theta_2(s_1 - \xi_1 + \xi_2) + \phi_2(s_1 - \xi_1 + \xi_2), \quad \forall s_1 \in [\xi_1, L_1] \quad (4) \quad 127$$

128 Lastly, the deformations ϕ_1 and ϕ_2 must allow the cylinders to
 129 form a close loop such that the isoperimetric constraints
 130 $\int_0^{L_1} \cos(\theta_1 + \phi_1) ds_1 = \int_0^{L_2} \cos(\theta_2 + \phi_2) ds_2 = 0$ are satisfied. In light of
 131 the compatibility condition in Eq. (4), this is equivalent to

$$\int_0^{\xi_1} \cos(\theta_1 + \phi_1) ds_1 = \int_0^{\xi_2} \cos(\theta_2 + \phi_2) ds_2 = - \int_{\xi_1}^{L_1} \cos(\theta_1 + \phi_1) ds_1 \quad (5) \quad 132$$

133 At this point it is convenient to define

$$\phi_a = \{\phi_1 : s_1 \in [\xi_1, L_1]\} \quad (6) \quad 135$$

136 This allows deformation to be represented by three independent
 137 functions ϕ_1 , ϕ_2 , and ϕ_a on the domains $[0, \xi_1]$, $[0, \xi_2]$, and
 138 $[\xi_1, L_1]$, respectively. By introducing ϕ_a , the boundary conditions
 139 reduce to

$$\phi_1(0) = \phi_2(0) = \phi_a(L_1) = 0 \quad (7) \quad 140$$

$$\phi_a(\xi_1) = \phi_1(\xi_1) = 2\pi + \phi_2(\xi_2) + \theta_2(\xi_2) - \theta_1(\xi_1) \quad (8) \quad 141$$

$$\int_0^{\xi_1} \cos(\theta_1 + \phi_1) ds_1 = \int_0^{\xi_2} \cos(\theta_2 + \phi_2) ds_2 = - \int_{\xi_1}^{L_1} \cos(\theta_1 + \phi_a) ds_1 \quad (9) \quad 142$$

143 It is important to note that these conditions explicitly prevent inter-
 144 penetration of the cylinders *only* along the contact zone (s_1
 145 $\in [\xi_1, L_1]$). 146

2.2 Energy Functional. The cylindrical walls are treated as 147
 148 inextensible *elastica*. Hence, extension and shear strains are ig-
 149 nored and the elastic strain energy is limited to bending. Let k_1
 150 and k_2 denote the dimensionless flexural rigidity of the bottom and
 151 top cylinders, respectively, where both k_i are normalized with re-
 152 spect to the flexural rigidity of cylinder 1, $D_1 = E_1 h_1 / 12(1 - \nu_1^2)$,
 153 with E_1 the elastic modulus, ν_1 Poisson's ratio, and h_1 the wall
 154 thickness. The total elastic strain energy of the system Γ can be

155 decomposed into the segments corresponding to the domains
 156 $[0, \xi_1]$, $[0, \xi_2]$, and $[\xi_1, L_1]$ as follows:

$$157 \quad \Gamma_1 = \int_0^{\xi_1} \frac{1}{2} k_1 \phi_{1,1}^2 ds_1, \quad \Gamma_2 = \int_0^{\xi_2} \frac{1}{2} k_2 \phi_{2,2}^2 ds_2$$

$$158 \quad \Gamma_3 = \int_{\xi_1}^{L_1} \left\{ \frac{1}{2} k_1 \phi_{a,1}^2 + \frac{1}{2} k_2 \left(\phi_{a,1} + \frac{1}{R_1} + \frac{1}{R_2} \right)^2 \right\} ds_1 \quad (10)$$

159 where $\phi_{i,j} = d\phi_i/ds_j$. The total potential energy of the system Π is
 160 computed by combining these elastic strain energies with the work
 161 U_f of the external load F , the virtual work U_λ of the isoperimetric
 162 constraints in Eq. (9), and the work of adhesion $W = \gamma a$. That is,

$$163 \quad \Pi = \Gamma_1 + \Gamma_2 + \Gamma_3 + U_f + U_\lambda - W \quad (11)$$

164 where

$$165 \quad U_f = \int_0^{\xi_1} F \sin(\theta_1 + \phi_1) ds_1 - \int_0^{\xi_2} F \sin(\theta_2 + \phi_2) ds_2 \quad (12)$$

166 and

$$167 \quad U_\lambda = \int_0^{\xi_1} \lambda_1 \cos(\theta_1 + \phi_1) ds_1 + \int_0^{\xi_2} \lambda_2 \cos(\theta_2 + \phi_2) ds_2$$

$$168 \quad + \int_{\xi_1}^{L_1} (\lambda_1 + \lambda_2) \cos(\theta_1 + \phi_a) ds_1 \quad (13)$$

169 The Lagrangian multipliers λ_1 and λ_2 in Eq. (13) are unknown
 170 constants and correspond to the internal ‘‘hoop’’ stress at the points
 171 $s_1 = s_2 = 0$. The total potential energy of the system may be ex-
 172 pressed by the functional

$$173 \quad \Pi = \int_0^{\xi_1} \left\{ \frac{1}{2} k_1 \phi_{1,1}^2 + F \sin(\theta_1 + \phi_1) + \lambda_1 \cos(\theta_1 + \phi_1) \right\} ds_1$$

$$174 \quad + \int_0^{\xi_2} \left\{ \frac{1}{2} k_2 \phi_{2,2}^2 - F \sin(\theta_2 + \phi_2) + \lambda_2 \cos(\theta_2 + \phi_2) \right\} ds_2$$

$$175 \quad + \int_{\xi_1}^{L_1} \left\{ \frac{1}{2} (k_1 + k_2) \phi_{a,1}^2 + k_2 \phi_{a,1} \left(\frac{1}{R_1} + \frac{1}{R_2} \right) + \frac{1}{2} k_2 \left(\frac{1}{R_1} + \frac{1}{R_2} \right)^2 \right.$$

$$176 \quad \left. + (\lambda_1 + \lambda_2) \cos(\theta_1 + \phi_a) - \gamma \right\} ds_1 \quad (14)$$

177 3 Analysis

178 At equilibrium, the energy functional Π must be stationary with
 179 respect to kinematically admissible variations of the form

$$180 \quad \phi_1 = \phi_1^* + \delta\phi_1, \quad \phi_2 = \phi_2^* + \delta\phi_2, \quad \phi_a = \phi_a^* + \delta\phi_a, \quad a = a^* + \delta a \quad (15)$$

181 Here, χ^* denotes the value of χ at equilibrium and $\delta\chi$ is an arbitrary
 182 but infinitesimally small variation from χ^* . In the subsequent
 183 analysis, it is convenient to define the Lagrangian densities

$$184 \quad \Lambda_1 = \frac{1}{2} k_1 \phi_{1,1}^2 + F \sin(\theta_1 + \phi_1) + \lambda_1 \cos(\theta_1 + \phi_1)$$

$$185 \quad \Lambda_2 = \frac{1}{2} k_2 \phi_{2,2}^2 - F \sin(\theta_2 + \phi_2) + \lambda_2 \cos(\theta_2 + \phi_2)$$

$$186 \quad \Lambda_a = \frac{1}{2} (k_1 + k_2) \phi_{a,1}^2 + k_2 \phi_{a,1} \left(\frac{1}{R_1} + \frac{1}{R_2} \right) + \frac{1}{2} k_2 \left(\frac{1}{R_1} + \frac{1}{R_2} \right)^2$$

$$187 \quad + (\lambda_1 + \lambda_2) \cos(\theta_1 + \phi_a) - \gamma \quad (16)$$

188 **3.1 Balance Laws.** Let $\delta\Pi_\phi$ denote the variation in Π in-
 189 duced by the first three variations in Eq. (15). Employing the

calculus of variations and noting that the variations must be kinematically admissible, it is straightforward to show that $\delta\Pi_\phi$ vanishes if and only if the balance laws

$$\frac{\partial \Lambda_1}{\partial \phi_1} - \frac{d}{ds_1} \left(\frac{\partial \Lambda_1}{\partial \phi_{1,1}} \right) = 0, \quad \frac{\partial \Lambda_2}{\partial \phi_2} - \frac{d}{ds_2} \left(\frac{\partial \Lambda_2}{\partial \phi_{2,2}} \right) = 0, \quad (17)$$

$$\frac{\partial \Lambda_a}{\partial \phi_a} - \frac{d}{ds_1} \left(\frac{\partial \Lambda_a}{\partial \phi_{a,1}} \right) = 0 \quad (18)$$

and natural boundary condition

$$\left(\frac{\partial \Lambda_1}{\partial \phi_{1,1}} \right)_{s_1=\xi_1} + \left(\frac{\partial \Lambda_2}{\partial \phi_{2,2}} \right)_{s_2=\xi_2} - \left(\frac{\partial \Lambda_a}{\partial \phi_{a,1}} \right)_{s_1=\xi_1} = 0 \quad (18)$$

are satisfied (see Appendix A for derivation). Equation (17) corresponds to the differential form of the moment balance along the segments $s_1 \in [0, \xi_1]$, $s_2 \in [0, \xi_2]$, and $s_1 \in [\xi_1, L_1]$, respectively, while Eq. (18) corresponds to the moment balance at the edge of the interface ($s_1 = \xi_1$).

Substituting the Lagrangian densities into Eq. (17) results in a system of three second-order ordinary differential equations. Solving these will introduce six constants of integration (c_1, c_2, \dots, c_6), resulting in altogether nine unknowns: $a, \lambda_1, \lambda_2, c_1, c_2, \dots, c_6$. However, so far, we have presented only eight linearly independent equations: the five boundary conditions in Eqs. (7) and (8), the two isoperimetric constraints in Eq. (9), and moment balance (18) at $s_1 = \xi_1$ and $s_2 = \xi_2$. In order to calculate the unknown constants, a ninth linearly independent equation is required. This is furnished by the fourth variation in Eq. (15) and is presented in Sec. 3.2.

3.2 Jump Condition. The fourth variation in Eq. (15) results in a variation of the potential energy that has the form $\delta\Pi_a = (d\Pi/da)\delta a$. Since δa is arbitrary, $\delta\Pi_a$ vanishes if and only if $d\Pi/da = 0$. Employing Leibniz’ integration rule, the chain rule, the balance laws in Eq. (17), and the natural boundary condition in Eq. (18), it follows that $d\Pi/da = 0$ reduces to

$$(\Lambda_a)_{s_1=\xi_1} - (\Lambda_1)_{s_1=\xi_1} - (\Lambda_2)_{s_2=\xi_2} + \left(\frac{\partial \Lambda_1}{\partial \phi_{1,1}} \right)_{s_1=\xi_1} \{ \phi_{1,1}(\xi_1) - \phi_{a,1}(\xi_1) \}$$

$$+ \left(\frac{\partial \Lambda_2}{\partial \phi_{2,2}} \right)_{s_2=\xi_2} \left\{ \phi_{2,2}(\xi_2) - \phi_{a,1}(\xi_1) - \frac{1}{R_1} - \frac{1}{R_2} \right\} = 0 \quad (19)$$

Details of the derivation are provided in Appendix B. Jump condition (19) provides the ninth equation necessary to complete the system of linear equations needed to solve for the nine unknown constants: $a, \lambda_1, \lambda_2, c_1, c_2, \dots, c_6$. Physically, Eq. (19) corresponds to the balance of the work of adhesion with the elastic energy release rate associated with variations of the arc length a of its value at equilibrium.

3.3 Solution. The governing equations are derived by substituting the expressions for Λ_1 , Λ_2 , and Λ_a into the above equations. A solution can easily be obtained by linearizing for small ϕ_1 and ϕ_2 . This yields the following set of governing equations (see Appendix C):

$$k_1 \phi_{1,11} = F \cos(\theta_1) - \lambda_1 \sin(\theta_1) \quad (20)$$

$$k_2 \phi_{2,22} = -F \cos(\theta_2) - \lambda_2 \sin(\theta_2) \quad (21)$$

$$(k_1 + k_2) \phi_{a,11} = -(\lambda_1 + \lambda_2) \sin(\theta_1) \quad (22)$$

Also, natural boundary condition (18) and jump condition (19) imply

$$k_1\{\phi_{1,1}(\xi_1) - \phi_{a,1}(\xi_1)\}k_2\left\{\phi_{2,2}(\xi_2) - \phi_{a,1}(\xi_1) - \frac{1}{R_1} - \frac{1}{R_2}\right\} = 0 \quad (23)$$

238 and

$$\frac{1}{2}k_1\{\phi_{1,1}(\xi_1) - \phi_{a,1}(\xi_1)\}^2 + \frac{1}{2}k_2\left\{\phi_{2,2}(\xi_2) - \phi_{a,1}(\xi_1) - \frac{1}{R_1} - \frac{1}{R_2}\right\}^2 = \gamma \quad (24)$$

242 respectively. As before, the kinematic boundary conditions are

$$\phi_1(0) = \phi_2(0) = \phi_a(L_1) = 0 \quad (25)$$

$$\phi_a(\xi_1) = \phi_1(\xi_1) = 2\pi + \phi_2(\xi_2) + \theta_2(\xi_2) - \theta_1(\xi_1) \quad (26)$$

245 Lastly, linearization reduces the isoperimetric constraints to

$$\int_0^{\xi_1} \{\cos(\theta_1) - \phi_1 \sin(\theta_1)\} ds_1 = \int_0^{\xi_2} \{\cos(\theta_2) - \phi_2 \sin(\theta_2)\} ds_2 = - \int_{\xi_1}^{L_1} \{\cos(\theta_1) - \phi_a \sin(\theta_1)\} ds_1 \quad (27)$$

248 Solving balance equations (20)–(22) yields six constants of integration, c_1, c_2, \dots, c_6 . Hence, there are altogether nine unknowns: $a, \lambda_1, \lambda_2, c_1, c_2, \dots, c_6$, to be determined by substituting the solutions to Eqs. (20)–(22) into Eqs. (23)–(27). Consequently, there is a system of nine equations with nine unknowns.

253 Numerical solutions to the system described in Eqs. (20)–(27) are presented in Figs. 3–5. In all three sets of figures, (a) depicts the deformation under a varying compressive load F , (b) the contact length a as a function of F , and (c) F as a function of stack height w . The results in (b) and (c) are provided for various values of the adhesion energy γ . Here, the compression distance, or the change in height of the stacked cylinders (equilibrium stack height minus the sum of undeformed cylinders) w , is defined as

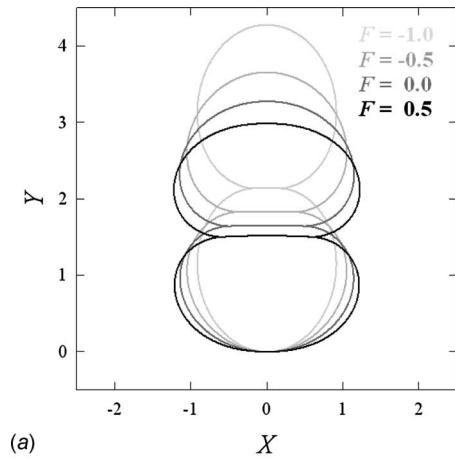
$$w = 2(R_1 + R_2) - \int_0^{\xi_1} \sin(\theta_1 + \phi_1) ds_1 + \int_0^{\xi_2} \sin(\theta_2 + \phi_2) ds_2 \quad (28)$$

261 The input parameters $(k_1, k_2, R_1, R_2, \gamma, F)$ and calculated values (a, w) are all unitless.

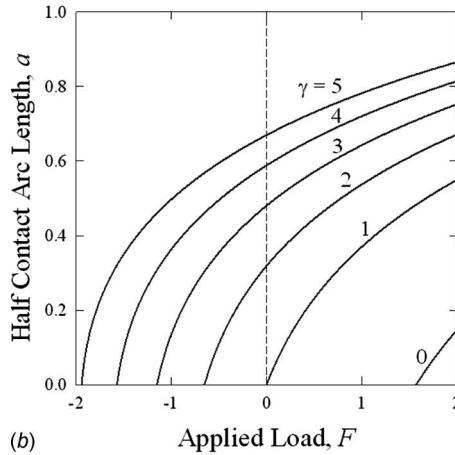
264 4 Worked Examples

265 The governing equations are derived using the principle of minimum potential energy. The potential energy functional comprises the strain energy created by elastic bending in both the contacting and noncontacting portions of the cylinders, the potential energy of the external load F , and the work of adhesion to expose new surfaces. Apart from the standard differential and boundary forms of moment balance (17) and (18), stationarity of the potential energy functional furnishes a jump condition at the edge ($s_1 = \xi_1, s_2 = \xi_2$) of contact zone (19). A simpler jump condition had previously been derived for adhesion of a single cylinder to a rigid, flat substrate, a result that has recently been shown to be equivalent to a discontinuity in the internal moment [31,35]. The jump condition in Eq. (19), however, has more terms since it concerns adhesion between two generally dissimilar thin-walled cylinders. Moreover, it does not appear to correspond to a discontinuity in internal moment and is instead related to a discontinuity in material (configurational) forces or Eshelbian energy-momentum.

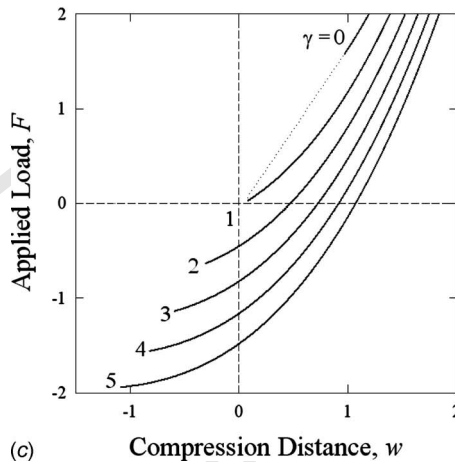
283 **4.1 Same Stiffness and Radii ($k_1 = k_2, R_1 = R_2$)**. Figure 3(a) shows the deformed cylinders with $k_1 = k_2 = 1$ and $R_1 = R_2 = 1$, under the coupled action of an external compressive load and adhe-



(a)



(b)



(c)

Fig. 3 Adhesion between two identical cylindrical shells with the same bending stiffness ($k_1 = k_2 = 1$) and radii ($R_1 = R_2 = 1$) under a compressive load F for $\gamma = 3$ (unless indicated otherwise). (a) Deformed profile with pole of bottom cylinder as reference. (b) Half contact arc length as a function of compressive load. (c) Change in stack height as a function of compressive load F and contact length a , with the dashed line indicating line contact ($a = 0$) where adhesion has no influence on the interacting cylinders.

286 sion with $\gamma = 3$. Both cylinders are flattened at their contact inter-
287 face and globally deformed to a pseudo-elliptic geometry, with the
288 lower pole of the bottom cylinder as the reference ($s_1 = 0$). The
289 deformation is symmetric with respect to the planar contact. In
290 this respect, the identical cylinders deform in a manner qualita-

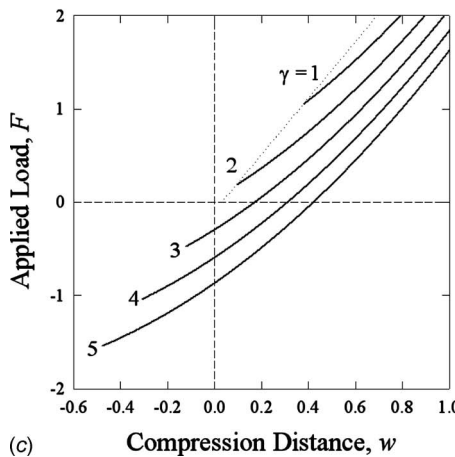
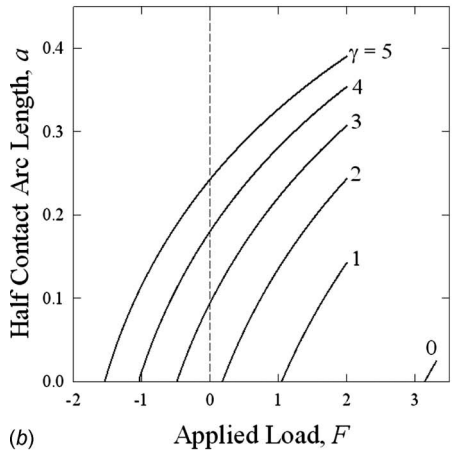
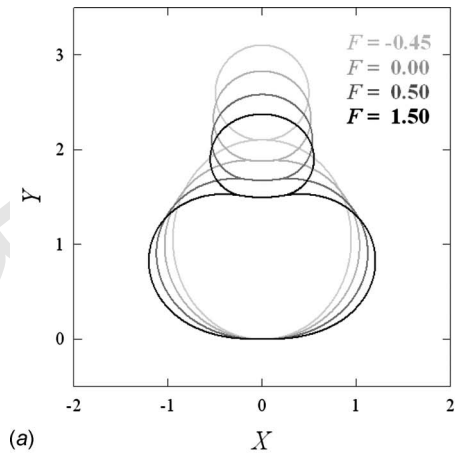


Fig. 4 Adhesion between two cylindrical shells with the same bending stiffness ($k_1=k_2=1$) but different radii ($R_1=1, R_2=0.5$) under a compressive load F for $\gamma=3$ (unless indicated otherwise). (a) Deformed profile. (b) Half contact arc length as a function of compressive load. (c) Change in stack height as a function of compressive load and contact length, with the dashed line indicating line contact ($a=0$) where adhesion has no influence.

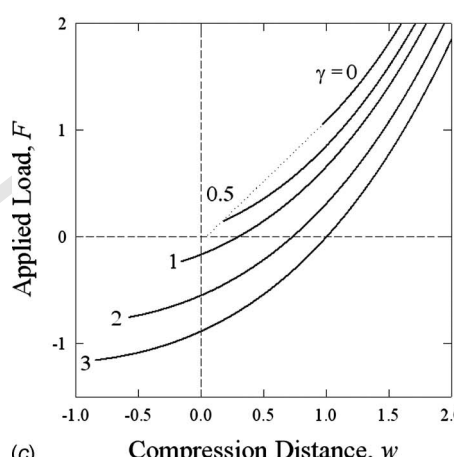
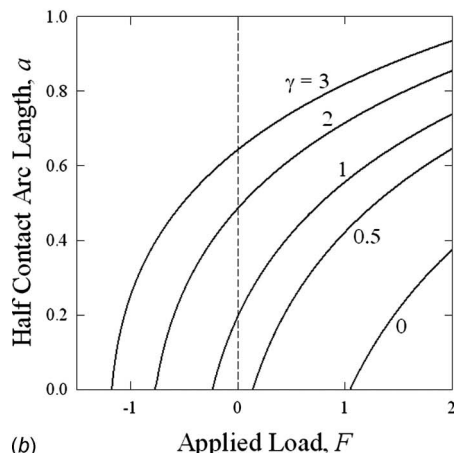
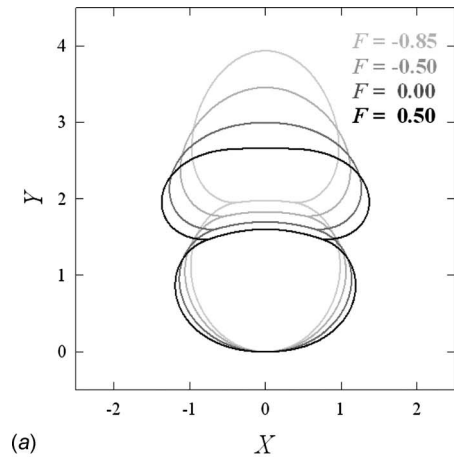


Fig. 5 Adhesion between two cylindrical shells with the same radii ($R_1=R_2=1$) but different bending stiffnesses ($k_1=1, k_2=0.5$) under a compressive load F for $\gamma=3$ (unless indicated otherwise). (a) Deformed profile. (b) Half arc contact length as a function of compressive load. (c) Change in stack height as a function of compressive load and contact length, with the dashed line indicating line contact ($a=0$) where adhesion has no influence.

291 tively similar (but not equivalent) to the adhesion of a single cylinder to a flat, rigid substrate. Figures 3(b) and 3(c) show the mechanical responses $a(F)$ and $F(w)$ for a range of γ . As F decreases, the contact shrinks and continues to be finite even when the external load turns tensile ($F < 0$). When the tensile load reaches the threshold, $F_0 = \min(F)$, the contact vanishes ($a=0$) and

the two adhering cylinders snap, leading to pull-off. The critical tensile load (negative F) increases with increasing γ ; e.g., $F_0(\gamma=3) \approx -1.1$ and $F_0(\gamma=5) = -2$. Interestingly, $F_0(\gamma=1) = 0$ is predicted, implying that the work of adhesion is insufficient to cause spontaneous adhesion of the two cylinders. A minimum compressive load is necessary to make finite contact ($a > 0$). In reality,

303 adhesion is the result of intersurface forces with *finite* range such
 304 that the cylinders interact even in the absence of intimate contact
 305 and a tensile load is always needed to separate the adherends. For
 306 $0 < \gamma < 1$, F_0 is positive at $a=0$ such that adhesion is irrelevant for
 307 $0 < F < F_0$ and II thus comprises the elastic deformation energy
 308 and potential energy due to external load only. Physically, when F
 309 falls below F_0 , the contact area remains a line ($a=0$) until w
 310 reduces to zero. There exists a minimal critical cylinder radius
 311 R_{min} , below which the contact is always zero (to be discussed in
 312 Sec. 5). In Fig. 3(c), the compression distance is always positive
 313 ($w > 0$) even in the absence of external load ($F=0$) as adhesion
 314 compels the two cylinders. As the load turns tensile ($F < 0$), w
 315 reduces further and the cylinder becomes more elongated about
 316 the vertical axis until pull-off occurs at the termini of all curves.

317 **4.2 Same Stiffness and Different Radii ($k_1=k_2, R_1=2R_2$) .**

318 Figure 4(a) shows two dissimilar cylinders with $k_1=k_2=1$ but
 319 $R_1=1$ and $R_2=0.5$. Here, the deformation about the curved contact
 320 becomes asymmetric. Elastic deformation is mainly confined to
 321 the larger cylinder even along the contact length. The relations
 322 $a(F)$ and $F(w)$ are similar to Figs. 3(b) and 3(c) despite a shift in
 323 F_0 . The $F(w)$ for $\gamma=1$ and 2 terminate as adhesion loses its influ-
 324 ence on the cylinders.

325 **4.3 Different Stiffness and Same Radii ($k_1=2k_2, R_1=R_2$) .**

326 Figure 5(a) shows dissimilar cylinders with $k_1=1$ and $k_2=0.5$, but
 327 $R_1=R_2=1$. The more compliant cylinder suffers from a larger de-
 328 gree of deformation. For $F=1$, the change in angle ϕ_2 is as large
 329 as 0.6 rad over much of the noncontacting portion. Hence, the
 330 small angle approximation used to derive Eqs. (20)–(22) and (27)
 331 is no longer suitable and the exact differential equation or higher
 332 order approximation is necessary. Moreover, because of the
 333 greater compliance, Figs. 5(b) and 5(c) are limited to $\gamma \leq 3$. For
 334 larger γ , the more compliant cylinder will spontaneously adhere to
 335 the stiffer cylinder and undergo deformation angles ϕ_2 that are
 336 well beyond the range of the small angle approximation.

337 **4.4 Example of Carbon Nanotubes.** A practical example is

338 the mechanical deformation of CNT in the presence of adhesion.
 339 Though the proper computation should incorporate the crystallo-
 340 graphic structure and orientation, we adopt the present continuum
 341 model and compare the results with molecular simulation by Tang
 342 et al. [22]. A comprehensive summary of published CNT materials
 343 parameters is given by Tu and Ou-Yang [25]. According to Sears
 344 and Batra [32], an equivalent elastic tube representation of CNT
 345 possesses sheet thickness $h=0.1$ nm, radius $R=0.6$ nm, and elas-
 346 tic modulus $E=3.0$ TPa. In order to achieve a contact arc length
 347 $a=0.1$ nm (17% of R), an adhesion energy of $\gamma \approx 1.0$ J m⁻² is
 348 required, which is a reasonable estimate of the van der Waals
 349 interactions. In the presence of water meniscus alone, γ
 350 ≈ 0.144 J m⁻², which falls below the critical adhesion energy γ^* ,
 351 the contact area is a line ($a=0$), and the corresponding pull-off
 352 load vanishes ($F_0=0$).

353 **5 Discussion**

354 It is worthwhile to compare the present model with the classical
 355 JKR and DMT models for adhering solid spheres. For example,
 356 the predicted pull-off force F_0 is found to have a much stronger
 357 dependence on the size (R) than stiffness (k), in reminiscence of
 358 $(F_0)_{JKR}=(3/2)\pi R\gamma$ and $(F_0)_{DMT}=\pi R\gamma$, where both depend only
 359 on the solid sphere dimension but not on materials stiffness. To
 360 make a more rigorous comparison, the pull-off force is normalized
 361 by $\pi R_{eff}\gamma$ with $R_{eff}^{-1}=R_1^{-1}+R_2^{-1}$ being the effective cylinder ra-
 362 dius. Figure 6(a) presents $R_1=R_2=1$ and $k_1=1$ for a range of k_2 .
 363 In the limit of large γ , $F_0/\pi R\gamma$ approaches an asymptote of ap-
 364 proximately 1/4, independent of k_1 and k_2 . On the other hand,
 365 once the adhesion energy falls below a threshold of γ^* , pull-off
 366 occurs at $F_0(\gamma \leq \gamma^*)=0$. Despite the similarity with JKR, cylindri-

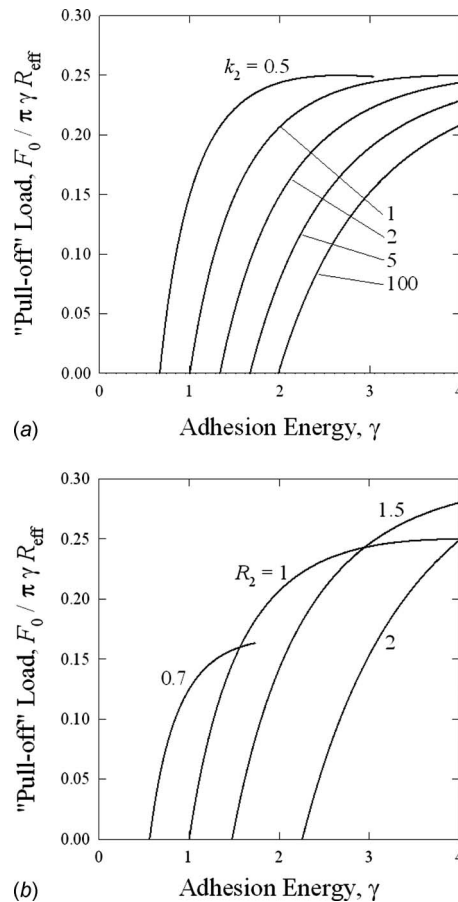


Fig. 6 Normalized pull-off strength F_0 as a function of adhesion energy γ . (a) Same radius ($R_1=R_2=1$), but different stiffness with $k_1=1$. (b) Same stiffness ($k_1=k_2=1$), but different radii with $R_1=1$.

cal shells deviate significantly at small γ . Figure 6(b) shows k_1 367
 $=k_2=1$ and $R_1=1$ for a range of R_2 . The monotonic increasing F_0 368
 again shows a minimum threshold with $F_0(\gamma \leq \gamma^*)=0$, but there 369
 does not exist a common threshold for large γ . The fact that the 370
 upper limit for each $F_0(\gamma)$ curve decreases with increasing R_2 371
 indicates that (i) the larger cylinder becomes more compliant and 372
 thus requires a smaller pull-off force, and (ii) the pull-off depends 373
 predominantly on the cylinder dimension alluding to the JKR 374
 model. 375

A fundamental difference between the current model and JKR 376
 is noted. A distinct feature of the JKR is the local deformation of 377
 the adhering spheres at the contact circle. In essence, the combin- 378
 ed applied load and adhesion force press a sphere against a 379
 rigid planar substrate to create a Hertz contact circle. While re- 380
 taining the contact circle, the adhesion force is then removed and 381
 replaced with a local deformation around the contact circle. This 382
 is done by assuming that a circular punch in full contact with a 383
 half elastic continuum pulls on the substrate giving rise to a linear 384
 “relaxation” and reduction in the approach displacement. Proper 385
 energy balance thus leads to a mechanical instability or pull-off at 386
 a critical tensile applied load and a nonzero contact radius. Should 387
 the essential relaxation be ignored, the contact circle always 388
 shrinks to zero (one-point contact) at pull-off. In general, the char- 389
 acteristic nonzero pull-off contact is expected in geometrically 390
 incompatible surfaces (e.g., spheres). Existing models for thin- 391
 walled vesicles ignore local deformation and indeed predict zero 392
 pull-off radius [33–35], though it must be emphasized that deter- 393
 mination of the exact pull-off radius proves to be quite elusive. 394
 Nevertheless, our present model does not consider local deforma- 395

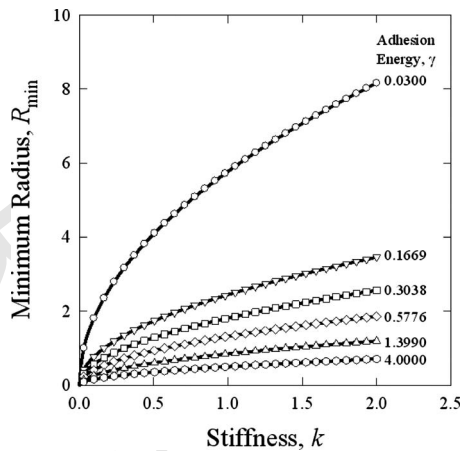


Fig. 7 Threshold radius for line contact ($a=0$) as function of stiffness and adhesion energy. ($R_{\min}=(k/\gamma)^{1/2}$ for $R_1=R_2=R$ and $k_1=k_2=k$.)

tion and the contact arc length must therefore reduce to zero (i.e., line contact) at pull-off. A comprehensive model is beyond the scope of this paper.

The eccentric behavior of line contact ($a=0$) present at small compressive external load (cf. Fig. 3(c)) is worth discussing. Based on the assumption that carbon nanotubes are planar graphene sheets folded into a cylindrical shell, Hui and co-workers [22,23] used an alternative method to derive a minimum cylindrical radius below which the adhesion contact remains a line: $R_{\min}=(k/\gamma)^{1/2}$ for $R_1=R_2=R$ and $k_1=k_2=k$. Compliant cylinders (small k) coupled with strong adhesion (large γ) is more prone to deformation and thus a small R_{\min} . The present model considers cylinders are initially stress-free. To deduce the relation between R_{\min} , k , and γ , values of R and k are randomly chosen, and the relation $a(F)$ is then found for a range of γ . The unique curve intersecting the origin ($a=0$ and $F=0$) corresponds to the value of $\gamma(R_{\min})$. For instance, in Fig. 3(b), $k=1$ and $R_{\min}=1$; therefore, $\gamma=1$ because the corresponding $a(F)$ intersects the origin. The numerical routine is repeated for a range of k and R combinations. Notwithstanding the distinctly different assumptions and analyses in the two models, an excellent comparison between our present model (data) and that of Hui and co-workers (curves) is shown in Fig. 7 for R_{\min} as a function of k for specific γ . The consistency is expected because no matter the cylinders possess an intrinsic stress, mechanical deformation to form the planar contact area causes a compressive stress to build up within the contact and immediately without, and thus raise the elastic energy of the system from the ground state of undeformed geometry. The current model is more general in the sense that dissimilar cylinders with different stiffnesses and dimensions are considered. Moreover, we deduce that the nonzero R_{\min} is a consequence of the global deformation of the cylinders and the local deformation within the contact arc length, instead of “a residual stress that increases the stiffness of smaller diameter tubes” [22].

Our present 2D cylindrical shell model sheds lights on the adhesion of 3D structures. One application is in cell aggregation, which is related to the formation and growth of natural, prosthetic, and malignant tissues. The existing model in literature treats cells as deformable solid spheres conforming to JKR theory. When these free entities come into contact due to thermal collision and vibration, interfacial adhesion occurs, followed by aggregation and coagulation. It is again emphasized that cells are not solid spheres but a viscoelastic cytoplasm encapsulated by a thin lipid bilayer membrane (shell). The present adhesion model properly addresses the nature of coupled shell deformation and adhesion, provides the constitutive relations between F , w , and a , and thus

yields the basis for a correct statistical portrayal of the Gibbs free energy and partition function of the grand canonical ensemble [36]. Immediate biomedical application is found in deriving the physical thresholds for cell aggregation (e.g., concentration, dimension, and temperature). Long-range surface forces can also be incorporated into the present model such that the adhering surfaces sense the presence of their counterpart even prior to direct contact, as shown in our latest work for freestanding membrane clamped at the periphery adhering to a planar substrate [20].

6 Conclusion

Using an energy balance, we derived the adhesion mechanics for two interacting elastic cylindrical shells with ranges of bending stiffness, radii, and adhesion energy. Relationships are established between the measurable quantities at equilibrium, applied load, stack height, contact length and deformed cylinder profiles, and the quasistatic adhesion-delamination trajectories. The graphs and trends presented have significant implications in the adhesion of similar and dissimilar interfaces in micro-/nanoshell structures. Such interactions are relevant to a variety of systems in nanoscience and technology, life-sciences, and tissue engineering.

Acknowledgment

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Appendix A

Following the definitions in Eqs. (14) and (16), the total potential energy of the system may be expressed as

$$\Pi = \int_0^{\xi_1} \Lambda_1 ds_1 + \int_0^{\xi_2} \Lambda_2 ds_2 + \int_{\xi_1}^{L_1} \Lambda_a ds_1 \quad (A1)$$

At equilibrium, Π must be stationary with respect to variations in ϕ_1 , ϕ_2 , and ϕ_a as well as their derivatives ($\phi_{1,1}$, $\phi_{2,2}$, and $\phi_{a,1}$). Applying these variations simultaneously to Π yields an expression of the form

$$\begin{aligned} \delta\Pi_\phi = & \int_0^{\xi_1} \left\{ \frac{\partial\Lambda_1}{\partial\phi_1} \delta\phi_1 + \frac{\partial\Lambda_1}{\partial\phi_{1,1}} \delta\phi_{1,1} \right\} ds_1 + \int_0^{\xi_2} \left\{ \frac{\partial\Lambda_2}{\partial\phi_2} \delta\phi_2 \right. \\ & \left. + \frac{\partial\Lambda_2}{\partial\phi_{2,2}} \delta\phi_{2,2} \right\} ds_2 + \int_{\xi_1}^{L_1} \left\{ \frac{\partial\Lambda_a}{\partial\phi_a} \delta\phi_a + \frac{\partial\Lambda_a}{\partial\phi_{a,1}} \delta\phi_{a,1} \right\} ds_1 \end{aligned} \quad (A2)$$

By the chain rule,

$$\begin{aligned} & \int_0^{\xi_1} \left\{ \frac{\partial\Lambda_1}{\partial\phi_{1,1}} \delta\phi_{1,1} \right\} ds_1 \\ & = \int_0^{\xi_1} \left\{ \frac{d}{ds} \left(\frac{\partial\Lambda_1}{\partial\phi_{1,1}} \delta\phi_1 \right) - \frac{d}{ds} \left(\frac{\partial\Lambda_1}{\partial\phi_{1,1}} \right) \delta\phi_1 \right\} ds_1 \\ & = \left(\frac{\partial\Lambda_1}{\partial\phi_{1,1}} \delta\phi_1 \right)_{s_1=\xi_1} - \left(\frac{\partial\Lambda_1}{\partial\phi_{1,1}} \delta\phi_1 \right)_{s_1=0} \\ & \quad - \int_0^{\xi_1} \left\{ \frac{d}{ds} \left(\frac{\partial\Lambda_1}{\partial\phi_{1,1}} \right) \delta\phi_1 \right\} ds_1 \end{aligned} \quad (A3)$$

Here, the operation $(f)_{x=y}$ denotes the value of f at $x=y$. Applying this same identity to the other integrals in Eq. (A2),

$$\begin{aligned} \delta\Pi_\phi = & \left(\frac{\partial\Lambda_1}{\partial\phi_{1,1}} \delta\phi_1 \right)_{s_1=\xi_1} - \left(\frac{\partial\Lambda_1}{\partial\phi_{1,1}} \delta\phi_1 \right)_{s_1=0} + \int_0^{\xi_1} \left\{ \frac{\partial\Lambda_1}{\partial\phi_1} \right. \\ & - \left. \frac{d}{ds} \left(\frac{\partial\Lambda_1}{\partial\phi_{1,1}} \right) \right\} \delta\phi_1 ds_1 + \left(\frac{\partial\Lambda_2}{\partial\phi_{2,2}} \delta\phi_2 \right)_{s_2=\xi_2} - \left(\frac{\partial\Lambda_2}{\partial\phi_{2,2}} \delta\phi_2 \right)_{s_2=0} \\ & + \int_0^{\xi_2} \left\{ \frac{\partial\Lambda_2}{\partial\phi_2} - \frac{d}{ds} \left(\frac{\partial\Lambda_2}{\partial\phi_{2,2}} \right) \right\} \delta\phi_2 ds_2 + \left(\frac{\partial\Lambda_a}{\partial\phi_{a,1}} \delta\phi_a \right)_{s_1=L_1} \\ & - \left(\frac{\partial\Lambda_a}{\partial\phi_{a,1}} \delta\phi_a \right)_{s_1=\xi_1} + \int_{\xi_1}^{L_1} \left\{ \frac{\partial\Lambda_a}{\partial\phi_a} - \frac{d}{ds} \left(\frac{\partial\Lambda_a}{\partial\phi_{a,1}} \right) \right\} \delta\phi_a ds_1 \end{aligned} \quad (A4)$$

At equilibrium, $\delta\Pi_\phi$ must vanish for kinematically admissible variations in ϕ_1 , ϕ_2 , and ϕ_a .

According to the boundary conditions in Eq. (7), both $\phi_1(0) = \phi_2(0) = \phi_a(L_1) = 0$ and

$$\phi_1(0) + \delta\phi_1(0) = \phi_2(0) + \delta\phi_2(0) = \phi_a(L_1) + \delta\phi_a(L_1) = 0$$

must be satisfied. Clearly, this implies $\delta\phi_1(0) = \delta\phi_2(0) = \delta\phi_a(L_1)$;

in other words, variations in the deflection ϕ_i must vanish at the points s_i where ϕ_i is prescribed. Similarly, the boundary conditions in Eq. (8) require that both the conditions

$$\phi_a(\xi_1) = \phi_1(\xi_1) = 2\pi + \phi_2(\xi_2) + \theta_2(\xi_2) - \theta_1(\xi_1) \quad (A6)$$

and

$$\begin{aligned} \phi_a(\xi_1) + \delta\phi_a(\xi_1) = & \phi_1(\xi_1) + \delta\phi_1(\xi_1) \\ = & 2\pi + \phi_2(\xi_2) + \delta\phi_2(\xi_2) + \theta_2(\xi_2) - \theta_1(\xi_1) \end{aligned} \quad (A7)$$

be satisfied. This implies $\delta\phi_a(\xi_1) = \delta\phi_1(\xi_1) = \delta\phi_2(\xi_2) = \delta\phi_\xi$. Substituting the boundary conditions expressions for $\delta\phi_i$ into Eq. (A4),

$$\begin{aligned} \delta\Pi_\phi = & \left\{ \left(\frac{\partial\Lambda_1}{\partial\phi_{1,1}} \right)_{s_1=\xi_1} + \left(\frac{\partial\Lambda_2}{\partial\phi_{2,2}} \right)_{s_2=\xi_2} - \left(\frac{\partial\Lambda_a}{\partial\phi_{a,1}} \right)_{s_1=\xi_1} \right\} \delta\phi_\xi \\ & + \int_0^{\xi_1} \left\{ \frac{\partial\Lambda_1}{\partial\phi_1} - \frac{d}{ds} \left(\frac{\partial\Lambda_1}{\partial\phi_{1,1}} \right) \right\} \delta\phi_1 ds_1 + \int_0^{\xi_2} \left\{ \frac{\partial\Lambda_2}{\partial\phi_2} \right. \\ & - \left. \frac{d}{ds} \left(\frac{\partial\Lambda_2}{\partial\phi_{2,2}} \right) \right\} \delta\phi_2 ds_2 + \int_{\xi_1}^{L_1} \left\{ \frac{\partial\Lambda_a}{\partial\phi_a} - \frac{d}{ds} \left(\frac{\partial\Lambda_a}{\partial\phi_{a,1}} \right) \right\} \delta\phi_a ds_1 \end{aligned} \quad (A8)$$

At this point, the variations $\delta\phi_\xi$, $\delta\phi_1$, $\delta\phi_2$, and $\delta\phi_a$ are all independent and arbitrary. Hence, in order for $\delta\Pi_\phi$ to vanish, the conditions

$$\begin{aligned} \frac{\partial\Lambda_1}{\partial\phi_1} - \frac{d}{ds_1} \left(\frac{\partial\Lambda_1}{\partial\phi_{1,1}} \right) = 0, \quad \frac{\partial\Lambda_2}{\partial\phi_2} - \frac{d}{ds_2} \left(\frac{\partial\Lambda_2}{\partial\phi_{2,2}} \right) = 0, \\ \frac{\partial\Lambda_a}{\partial\phi_a} - \frac{d}{ds_1} \left(\frac{\partial\Lambda_a}{\partial\phi_{a,1}} \right) = 0 \end{aligned} \quad (A9)$$

$$\left(\frac{\partial\Lambda_1}{\partial\phi_{1,1}} \right)_{s_1=\xi_1} + \left(\frac{\partial\Lambda_2}{\partial\phi_{2,2}} \right)_{s_2=\xi_2} - \left(\frac{\partial\Lambda_a}{\partial\phi_{a,1}} \right)_{s_1=\xi_1} = 0 \quad (A10)$$

must be satisfied. It is important to note that boundary condition (A10) results from simultaneously applying the first three variations in Eq. (15). This is necessary since the variations are not independent, but related through the boundary conditions (7) and (8). Failure to incorporate these conditions into the calculus of variations would either eliminate kinematic constraints or introduce nonexistent ones.

Appendix B

At equilibrium, Π must be stationary with respect to variations in a . Since δa is arbitrary, the corresponding variation in Π , $\delta\Pi_a = (d\Pi/da)\delta a$, vanishes if and only if $d\Pi/da = 0$. Since both ξ_1 and ξ_2 depend on a , $d\Pi/da$ must be evaluated using Leibniz' integration rule

$$\begin{aligned} \frac{d\Pi}{da} = & (\Lambda_1)_{s_1=\xi_1} \frac{d\xi_1}{da} + (\Lambda_2)_{s_2=\xi_2} \frac{d\xi_2}{da} - (\Lambda_a)_{s_1=\xi_1} \frac{d\xi_1}{da} + \int_0^{\xi_1} \left\{ \frac{\partial\Lambda_1}{\partial\phi_1} \frac{d\phi_1}{da} \right. \\ & + \left. \frac{\partial\Lambda_1}{\partial\phi_{1,1}} \frac{d\phi_{1,1}}{da} \right\} ds_1 + \int_0^{\xi_2} \left\{ \frac{\partial\Lambda_2}{\partial\phi_2} \frac{d\phi_2}{da} + \frac{\partial\Lambda_2}{\partial\phi_{2,2}} \frac{d\phi_{2,2}}{da} \right\} ds_2 \\ & + \int_{\xi_1}^{L_1} \left\{ \frac{\partial\Lambda_a}{\partial\phi_a} \frac{d\phi_a}{da} + \frac{\partial\Lambda_a}{\partial\phi_{a,1}} \frac{d\phi_{a,1}}{da} \right\} ds_1 \end{aligned} \quad (B1)$$

In light of the balance law (17),

$$\begin{aligned} \int_0^{\xi_1} \left\{ \frac{\partial\Lambda_1}{\partial\phi_1} \frac{d\phi_1}{da} + \frac{\partial\Lambda_1}{\partial\phi_{1,1}} \frac{d\phi_{1,1}}{da} \right\} ds_1 \\ = \int_0^{\xi_1} \left\{ \frac{d}{ds_1} \left(\frac{\partial\Lambda_1}{\partial\phi_{1,1}} \right) \frac{d\phi_1}{da} + \frac{\partial\Lambda_1}{\partial\phi_{1,1}} \frac{d\phi_{1,1}}{da} \right\} ds_1 \\ = \int_0^{\xi_1} \frac{d}{ds_1} \left\{ \frac{\partial\Lambda_1}{\partial\phi_{1,1}} \frac{d\phi_1}{da} \right\} ds_1 \\ = \left(\frac{\partial\Lambda_1}{\partial\phi_{1,1}} \frac{d\phi_1}{da} \right)_{s_1=\xi_1} - \left(\frac{\partial\Lambda_1}{\partial\phi_{1,1}} \frac{d\phi_1}{da} \right)_{s_1=0} \end{aligned} \quad (B2)$$

Applying this identity to the other two integrals in Eq. (B1) and noting that $d\xi_1/da = d\xi_2/da = -1$,

$$\begin{aligned} \frac{d\Pi}{da} = & (\Lambda_a)_{s_1=\xi_1} - (\Lambda_1)_{s_1=\xi_1} - (\Lambda_2)_{s_2=\xi_2} + \left(\frac{\partial\Lambda_1}{\partial\phi_{1,1}} \frac{d\phi_1}{da} \right)_{s_1=\xi_1} \\ & - \left(\frac{\partial\Lambda_1}{\partial\phi_{1,1}} \frac{d\phi_1}{da} \right)_{s_1=0} + \left(\frac{\partial\Lambda_2}{\partial\phi_{2,2}} \frac{d\phi_2}{da} \right)_{s_2=\xi_2} - \left(\frac{\partial\Lambda_2}{\partial\phi_{2,2}} \frac{d\phi_2}{da} \right)_{s_2=0} \\ & + \left(\frac{\partial\Lambda_a}{\partial\phi_{a,1}} \frac{d\phi_a}{da} \right)_{s_1=L} - \left(\frac{\partial\Lambda_a}{\partial\phi_{a,1}} \frac{d\phi_a}{da} \right)_{s_1=\xi_1} \end{aligned} \quad (B3)$$

Next, by natural boundary condition (18), $d\Pi/da$ reduces to

$$\begin{aligned} \frac{d\Pi}{da} = & (\Lambda_a)_{s_1=\xi_1} - (\Lambda_1)_{s_1=\xi_1} - (\Lambda_2)_{s_2=\xi_2} \\ & + \left\{ \frac{\partial\Lambda_1}{\partial\phi_{1,1}} \left(\frac{d\phi_1}{da} - \frac{d\phi_a}{da} \right) \right\}_{s_1=\xi_1} - \left(\frac{\partial\Lambda_1}{\partial\phi_{1,1}} \frac{d\phi_1}{da} \right)_{s_1=0} \\ & + \left(\frac{\partial\Lambda_2}{\partial\phi_{2,2}} \right)_{s_2=\xi_2} \left\{ \left(\frac{d\phi_2}{da} \right)_{s_2=\xi_2} - \left(\frac{d\phi_a}{da} \right)_{s_1=\xi_1} \right\} \\ & - \left(\frac{\partial\Lambda_2}{\partial\phi_{2,2}} \frac{d\phi_2}{da} \right)_{s_2=0} + \left(\frac{\partial\Lambda_a}{\partial\phi_{a,1}} \frac{d\phi_a}{da} \right)_{s_1=L_1} \end{aligned} \quad (B4)$$

Equation (B4) represents the general expression for $d\Pi/da$ at equilibrium. It can be further simplified by applying the boundary conditions in Eqs. (7) and (8), which restrict not only ϕ_1 , ϕ_2 , and ϕ_a , but also their derivatives with respect to a . According to Eq. (7), ϕ_1 , ϕ_2 , and ϕ_a are all prescribed at $s_1=0$, $s_2=0$, and $s_1=L_1$, respectively. Since these conditions must hold for all values of a , the derivatives $(d\phi_1/da)_{s_1=0}$, $(d\phi_2/da)_{s_2=0}$, and $(d\phi_a/da)_{s_1=L_1}$ must equal zero. This is equivalent to the condition in the calculus of variations that the variation of prescribed end points must vanish.

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555 In contrast, $(d\phi_1/da)_{s_1=\xi_1}$, $(d\phi_2/da)_{s_2=\xi_2}$, and $(d\phi_a/da)_{s_1=\xi_1}$ are
 556 nonzero and must be computed using the boundary conditions in
 557 Eq. (8). According to Eq. (15) and the fundamental theorem of
 558 calculus,

$$559 \quad \phi_1(\xi_1) = \phi_1^*(\xi_1) + \delta\phi_1(\xi_1) = \phi_1^*(\xi_1^*) - \delta a \phi_{a,1}^*(\xi_1^*) + \delta\phi_1(\xi_1) \\ 560 \quad + O(\delta a^2) \quad (B5)$$

561 where $\xi_1^* = L_1 - a^*$ and a^* is the value of a at equilibrium. Simi-
 562 larly,

$$563 \quad \phi_a(\xi_1) = \phi_a^*(\xi_1^*) - \delta a \phi_{a,1}^*(\xi_1^*) + \delta\phi_a(\xi_1) + O(\delta a^2) \quad (B6)$$

564 which, according to boundary condition (8), must be equivalent to
 565 $\phi_1(\xi_1)$. Since δa is infinitesimally small, terms of order $O(\delta a^2)$
 566 may be omitted and so the conditions $\phi_1(\xi_1) = \phi_a(\xi_1)$ and
 567 $\phi_1(\xi_1^*) = \phi_a(\xi_1^*)$ together imply

$$568 \quad -\delta a \phi_{a,1}^*(\xi_1^*) + \delta\phi_1(\xi_1) = -\delta a \phi_{a,1}^*(\xi_1^*) + \delta\phi_a(\xi_1) \quad (B7)$$

569 Dividing both sides by δa , taking the limit as $\delta a \rightarrow 0$, and then
 570 rearranging terms,

$$\left(\frac{d\phi_a}{da}\right)_{s_1=\xi_1} - \left(\frac{d\phi_1}{da}\right)_{s_1=\xi_1} = \phi_{a,1}(\xi_1) - \phi_{1,1}(\xi_1) \quad (B8)$$

571 where the asterisk denoting the value at equilibrium is henceforth
 572 omitted. Using the same argument for $\phi_2(\xi_2)$, it follows from Eq.
 573 (8) that

$$\left(\frac{d\phi_a}{da}\right)_{s_1=\xi_1} - \left(\frac{d\phi_2}{da}\right)_{s_2=\xi_2} = \phi_{a,1}(\xi_1) - \phi_{2,2}(\xi_2) + \frac{1}{R_1} + \frac{1}{R_2} \quad (B9)$$

574 Substituting this into Eq. (19) and setting $d\Pi/da$ equal to zero
 575 yield the following jump condition at equilibrium:

$$576 \quad (\Lambda_a)_{s_1=\xi_1} - (\Lambda_1)_{s_1=\xi_1} - (\Lambda_2)_{s_2=\xi_2} + \left(\frac{\partial\Lambda_1}{\partial\phi_{1,1}}\right)_{s_1=\xi_1} \{\phi_{1,1}(\xi_1) - \phi_{a,1}(\xi_1)\} \\ 577 \quad + \left(\frac{\partial\Lambda_2}{\partial\phi_{2,2}}\right)_{s_2=\xi_2} \left\{ \phi_{2,2}(\xi_2) - \phi_{a,1}(\xi_1) - \frac{1}{R_1} - \frac{1}{R_2} \right\} = 0 \quad (B10)$$

580 Appendix C

581 Assuming that the deflections ϕ_1 , ϕ_2 , and ϕ_a are small, the
 582 Lagrangian densities may be approximated as

$$583 \quad \Lambda_1 = \frac{1}{2}k_1\phi_{1,1}^2 + F\{\sin(\theta_1) + \phi_1 \cos(\theta_1)\} + \lambda_1\{\cos(\theta_1) - \phi_1 \sin(\theta_1)\}$$

$$584 \quad \Lambda_2 = \frac{1}{2}k_2\phi_{2,2}^2 - F\{\sin(\theta_2) + \phi_2 \cos(\theta_2)\} + \lambda_2\{\cos(\theta_2) - \phi_2 \sin(\theta_2)\}$$

$$585 \quad \Lambda_a = \frac{1}{2}(k_1 + k_2)\phi_{a,1}^2 + k_2\phi_{a,1}\left(\frac{1}{R_1} + \frac{1}{R_2}\right) + \frac{1}{2}k_2\left(\frac{1}{R_1} + \frac{1}{R_2}\right)^2 \\ 586 \quad + (\lambda_1 + \lambda_2)\{\cos(\theta_1) - \phi_a \sin(\theta_1)\} - \gamma \quad (C1)$$

587 The balance equations are obtained by substituting these expres-
 588 sions into the Euler-Lagrange differential equation (17). This
 589 yields

$$590 \quad k_1\phi_{1,11} = F \cos(\theta_1) - \lambda_1 \sin(\theta_1) \quad (C2)$$

$$591 \quad k_2\phi_{2,22} = -F \cos(\theta_2) - \lambda_2 \sin(\theta_2) \quad (C3)$$

$$592 \quad (k_1 + k_2)\phi_{a,11} = -(\lambda_1 + \lambda_2)\sin(\theta_1) \quad (C4)$$

593 where $\phi_{i,jj} = d^2\phi_i/ds_j^2$.

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